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AFRPL-TR-66-92

R-6354-2

(Unclassified Title)

INHIBITED N2O4

Second Quarterly Progress Report

Composition Research Unit Chemistry Section Research Division

Rocketdyne
A Division of North American Aviation

April 1966

Group 4
Downgraded at 3-Year Intervals
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R-6354-2

(Unclassified Title)

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Second Quarterly Progress Report

Composition Research Unit Chemistry Section Research Division

Rocketdyne A Division of North American Aviation

April 1966

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### **FOREWORD**

The manager reported herein was supported by the air Force Rocket Propulsion Laboratory, Research and Technology Division, Edwards Air Force Base, California, Air Force Systems Command, United States Air Force, under contract number AF04(611)10809, project number 1148, BPSN 623148, program structure number 750G, with Ralph Fargnoli, 2nd/Lt/USAF, RPCL, serving as Froject Monitor. This quarterly technical progress report was prepared under Rocketdyne G.O. 8728 in compliance with Part I, Paragraph B, and line item 6 of DD1423, and IL-STD-847 (USAF).

The work described covers the period 1 October 1965 through 31 March 1966. The Responsible Scientist for this program is Dr. Hubert E. Dubb of the Analytical Chemistry Group, headed by Dr. B. L. Tuffly. was carried out by members of the Composition Research Unit supervised by Dr. V. H. Dayan, the Chemical Sensor Research Unit supervised by Mr. I. Lysyj, the Santa Susana Field Laboratory Chemical Analysis Unit supervised by Miss N. V. Sutton, the Measurement Systems Unit supervised by Mr. R. E. Nelson, the Synthetic Chemistry Group headed by Dr. E. A. Lawton, and the Propellant Engineering Group, headed by Dr. J. J. Kalvinskas.

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### ABSTRACT

This program was originally concerned with evaluating a new storable liquid oxidizer, INTO, which is NTO inhibited with a fluorine-containing oxidizer,  $F_3NO$ .

Storability results have shown that  $F_3^{N0}$  is not compatible with NTO at elevated temperatures. FNO $_2$  has been substituted as the inhibiting fluorine oxidizer and this new propellant system is called INTO-2. Its constituents have been found to be compatible at 70 C.

Long- and short-term corrosivity tests have been initiated. The 7-day compatibility tests with non-metallics have been completed in NTO, INTO-1, and INTO-2. Results indicate that INTO-2 is more compatible than INTO-1 with the materials studied.

There was little difference between NTO and INTO-2.

Conductance measurements were completed on INTO-1. Results indicated that galvanic corrosion will not be a problem.

INTO-2 has been prepared by reaction of fluorine with NTO and the resulting mixture has been shown to react with added water to bring the water equivalent down to 0.03 percent.

Analytical methods have been developed for the analysis of INTO-1 and INTO-2.

(Confidential Abstract)

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### INTROL CTION

The use of nitrogen tetrczide (NTO), the most widely used storable liquid oxidizer in the United States, has been continually hampered by corrosion problems. Dry NTO is not a highly corrosive liquid, but moist NTO is extremely corrosive because of the formation of nitric and nitrous acids by the reaction of NTO with water. The present program was undertaken with the objective of determining whether an oxidizer system consisting of NTO with a small amount of  $F_3$ NO added would be less corrosive than moist NTO.  $F_3$ NO was chosen as an additive because it had been demonstrated by nuclear magnetic resonance (n.m.r.) spectroscopy and by Turner bulb water equivalency measurements that this compound reacted with the nitric and nitrous acids present in moist NTO to form HF and to greatly reduce the nitric and nitrous acid content of the propellant. The other reaction anticipated was the one between  $O_2$  produced by the  $F_3$ NO-acid reaction and the NO present in NTO, the product of which was NTO. Thus, no new harmful species were formed.

The technical position at the start of the program was then one of knowledge that (1) dry NTO is not highly corrosive, (2) wet NTO is highly corrosive, and (3) adding  $F_{\overline{3}}$ NO to wet NTO reduces the nitric and nitrous acid content of the propellant with the concurrent production of HF. In addition, n.m.r. spectra taken over a 1-week period indicated no significant decomposition of the  $F_{\overline{3}}$ NO-NTO system.

As a result of compatibility tests made since the start of the program, it is now known that the  $F_5NO-NTO$  system is not storable at elevated (53 and 70 C) temperatures because the constituents react to form  $FNO_2$ .  $FNO_2$  has therefore been substituted for  $F_5NO$  as the additive. The compatibility of  $FNO_2$  with NTO has been demonstrated at 53 and 70 C. Also, it has been shown that  $FNO_2$  is as effective as  $F_5NO$  in reducing the water content of moist NTO.

As a consequence of the experimental results summarized above, the objective of the program has been changed from evaluating the operational capability of INTO ( $F_{\tau}NO + NTO$ ) to evaluating the operational capability

of INTO-2 (FNO<sub>2</sub> + NTO). The revised objectives of the program are to make an engineering evaluation of the corrosion properties and storability of INTO-2, to determine some of its basic physical and chemical properties, and to develop methods for its chemical analysis.

PHASE I: ENGINEERING EVALUATION

PHYSICAL TESTING

### Corrosivity Tests

The tests and evaluations applicable to INTO-1 (NTO +  $F_3$ NO), as defined in the first quarterly report, are also applicable to INTO-2 (NTO +  $FNO_2$ ). The list of materials will be repeated because of some minor changes from the original list (Table 1).

TABLE 1

# MATERIALS TO BE SUBJECTED TO CORROSIVITY TESTING

Group I Iron Alloys	Group II Aluminum Alloys	Group III Nonmetals
304 Stainless Steel	ь061-Т6	Kel-F
316 Stainless Steel	7075 <b>-</b> I73	Teflon
321 Stainless Steel	Tens 50	Viton A
4400 Stainless Steel	2219 <b>-T</b> 71	Kynar
AM 350 SCT	2014 <b>-</b> T6	Phenolic Resin-Cured
1018 Carbon Steel	2024	Butyl Rubber
Weld Samples	Weld Samples	

In the Group I iron alloys, 321 stainless steel was added because it is a representative, weldable, stainless steel and does not require the special annealing which is required by 316 stainless steel to reduce oxidization at the weld.

For the Group II aluminum alloys, the 2024 alloy was substituted for the 2219 alloy which was unavailable at this time but should be available approximately 1 May 1966. Because the 2024 alloy has chemical, physical

and corrosion resistance properties quite similar to the 2219 alloy, it was substituted in order not to delay the balance of the testing. A comparison of the chemical properties is shown in Table 2. The 2024 and 7075 aluminum alloys are considered "not weldable" because at this time they are only weldable under controlled laboratory conditions and not normally welded in production. As a result, only nonwelded specimens of the 2024 and 7075 alloys will be tested.

Viton A, military specification MIL-R-25897 Class I, has been added to the Group III nonmetals, because of its possible use in NTO systems as an O-ring material.

Because the investigation is comparative in nature, and its objective is to evaluate the effectiveness of the additive in preventing corrosion, specimens will be exposed to five propellant compositions. These are:

- 1. NTO (military specification, 0.10 weight percent H<sub>2</sub>0 equivalent and a minimum 99.5 percent assay)
- 2. NTO + 0.2 weight percent  $H_0$ 0
- 3. NTO (military specification) + 2.0 weight percent  ${
  m FNO}_2$
- 4. NTO + 0.2 weight percent  $H_2^0$  + 2.0 weight percent  $FNO_2$
- 5. NTO + 0.5 weight percent HF

All specimens were cleaned using a soap solution followed by a water rinse, followed by an acetone rinse. The specimens were then dried and weighed to the nearest 0.10 milligram. All weights were recorded as shown in the sample form presented as Table 3.

A coding system was established to define each specimen and its test conditions and is delineated in Table 4. Letters were assigned for each material, numbers were assigned for each test condition, and Roman numerals were assigned for each propellant composition.

The test bombs, valves, and fittings were vapor degreased, rinsed with trichloroethylene and then rinsed with acetone. The valves were completely disassembled for cleaning. Fluorolule was used as a lubricant on the valve stem assemblies.

TABLE 2

COMPARISON OF CHEMICAL COMPOSITION

						*	lloy E	Alloy Elements, percent	ercent			
Si Fe	Fe		Cu	Mn	Cu Mn Mg Cr Zn Ti	$\mathbf{cr}$	Zn	Ti	Λ	Zr	0ther	Įγ
0.5-1.2 1.0	1.0		3.9-5.0	0.4-1.2	.9-5.0 0.4-1.2 0.2-0.8 0.1 0.25 0.15	0.1	0.25	0.15			0.05 (0.15) Remainder	Renainder
0.5 0.5 3	0.5		3.8-4.9	0.4-1.2	3.8-4.9 0.4-1.2 0.05 0.1 0.1	0.1	0.1		ı	l	0.05 (0.15) Remainder	Remainder
0.2 0.3 5.	0.3		5.8-6.8	0.2-0.4	0.03	- 1	0.1	0.02-0.1	0.05-0.15	0.10-0.25	.8-6.8 0.2-0.4 0.02 — 0.1 0.02-0.1 0.05-0.15 0.10-0.25 0.05 (0.15) Remainder	Remainder
_	_	_	-			_	_	_	_	_		

TABLE 3
RECORDING WEIGHTS

	MA	TERIAL: 304 L Stai	nless Steel,Cod	Г	A, With	
	Code	Starting Weight, grams	Weight After, grams	Weight Change, gm	Сотр	Comments and Metalurgical Observations
	A-l	2.5954			I	
	A-5	2.6451			II	
	<b>A-</b> 9	2.5663	•		III	
L	A-13	2.8196		Ì	IV	
I	A-17	2.0594			v	
<b>Q</b> U	A-21	2.8311			I	
I	A-25	2.5726			II	
D	A-29	2.3024			III	
P H	A-33	2.8763			IV	
A	A-37	2.1428			v	
S E	A-41	2.1450			I	
E	A-45	2.6206		}	II	
	A-49	2.5269			III	
	A-53	2.8019			IV	
	<b>A-</b> 57	2.4270			v	
	<b>A</b> -3	2.1296			I	
	<b>A-</b> 7	2.7431			II	
	A-11	2.7722			III	
37	<b>A-</b> 15	2.1421			IV	Small metal whiske on specimen on edg
V A	<b>A-19</b>	2.5568			v	
<b>P</b> 0	A-23	2.1913			I	
R	A-27	2.8816			II	
P	A-31	2.9798			III	
H A	A-35	2.3819			IV	
S E	A-39	2.6932			v	1
E	A-43	2.4086			I	1
	A-47	2.6352			II	
	<b>A-</b> 51	2.9236			III	
	A-55	2.3527			IV	
	<b>A</b> -59	2.6285		}	v	

TABLE 4

### SPECIMEN ALPHANUMERIC CODING SYSTEM

### Alphabetical Connotation

Iron <u>Alloys</u>	Code <u>Letter</u>	Aluminum Alloys	Code <u>Letter</u>	Nonmetals	Code <u>Letter</u>
304 Stainless Steel	A	6061-Т6	G	Kel-F	N
321 Stainless Steel	В	70 <b>7</b> 5- <b>T</b> 73	H	Kynar	0
440 C Stainless Steel	C	Tens-50-T6	I	Butyl rubber	P
AM 350 SCT	D	2219- <b>T</b> 71	J	Viton "A"	Q
1018 Carbon Steel	${f E}$	2014-T6	K	Teflon	R
316 Stainless Steel	${f F}$	2024-T6	${f L}$		

### Numerical Connotation

<u>Circumstance</u>	Numerical Notation
Liquid Phase, Welded Liquid Fhase, No Weld	1, 5, 9, 13,57 2, 6, 10, 14,58
Vapor Phase, Welded Vapor Phase, No Weld	3, 7, 11, 15,59 4, 8, 12, 16,60
30-Day Test at Ambie. † Temperature	1, 2, 3, 4, ——20 21, 22, 23, ——40
20-Month Test at Ambient Temperature	22, 22, 23,
30-Day Test at 70 C	41, 42, 43,60

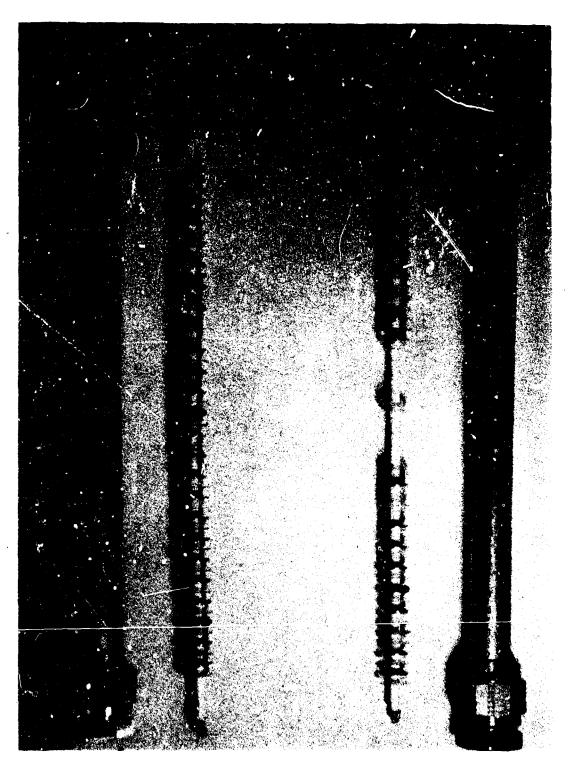
### Propellant Composition Designation

Composition	Roman Numeral
NTO (MIL Spec 0.10 weight percent H <sub>2</sub> 0 equivalent and a minimum 99.5 percent assay) <sup>2</sup>	I
NTO + 0.2 Weight Percent H <sub>0</sub> 0	II
N.O (MIL Spec)+ 2.0 weight percent FNO	III
NTO + 0.2 Weight Percent H <sub>0</sub> 0 + 2.0 Weight Percent FNO <sub>2</sub>	IV
NTO + 0.5 Weight Percent HF	v

Test specimens were assembled on rods using Teflon spacers and Kel-F separators between each specimen (Fig.  $^{1}$ ). A detailed description was presented in the first quarterly report.

The test bomb is loaded on to a vacuum rack (Fig. 2). The space between valves No. 1 and 2 is evacuated, valve No. 2 is closed, and valve No. 1 is opened. With valves No. 3 through 5 open, the test bomb and loading assembly are evacuated. Valve No. 4 is closed and the loading assembly is pumped down to a high vacuum. Valve No. 3 is closed and valve No. 4 is opened, so that the test bomb is pumped down to a high vacuum. At this time the dewar flask is filled with liquid nitrogen. Valve No. 2 is opened, and the loading assembly is allowed to fill for 5 minutes. After the loading assembly is filled, valves No. 2 and 5 are closed isolating the loading system from the vacuum system and propellant reservoir. Valve No. 3 is then opened and the transfer is carried out for a given period of time (approximately 10 minutes). During the transfer, the test bomb is kept at liquid nitrogen temperature. Upon completion of the transfer, valves No. 3 and 4 are closed and the bomb is removed from the loading system. The bomb is allowed to warm and then is weighted to verify transfer of a 65-milliliter volume. Smaller bombs and a smaller loading assembly are used for the nonmetal samples.

Evaluation of the nonmetals compatibility tests was based on weight change (when feasible) and visual examination for swelling, dissolution, cracking, etc. Seven-day tests of the compatibility of INTO-1 and NTO with Kel-F, Teflon, Kynar, Viton A, and Butyl rubber have been completed. Identical specimens were immersed in the liquid and in the vapor phase at ambient temperature. The first three materials appeared unaffected except for color changes. The Viton A cracked in both NTO and INTO-1 and Butyl rubber was completely dissolved in INTO-1 and seemingly unaffected by NTO. The F<sub>3</sub>NO in the INTO was not completely destroyed by the reaction with the Butyl rubber. The immersed and vapor phase specimens seemed to be identically affected during both the INTO-1 and NTO tests.



Fe Alloys

Al Alloys

Figure 1. 30-Day and 20-Month Test Specimens and Bombs

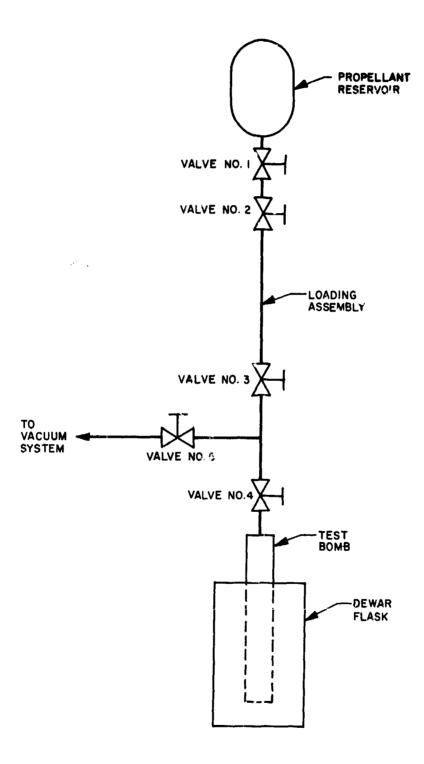


Figure 2. Loading Schematic

Seven-day compatibility tests of the nonmetal specimens have been repeated with NTO (military specification) and INTO-2, both immersed in the liquid and in contact with the vapor phase at ambient temperature.

The first three materials (Kel-F, Teflon, and Kynar) appeared unaffected by either NTO or INTO-2 except for color change and a very slight surface softness. The Viton A exposed to both NTO and INTO-2 exhibited extreme swelling (nearly twice the original size) as observed upon removal from the test bombs. After exposure to atmosphere for 30 minutes, the specimens reduced in size to approximately one and one-tenth the original size. There was some splitting around the inside diameter of the specimens exposed to the liquid phase of both the NTO and INTO-2. The Viton A specimen exposed to liquid phase INTO-2 had approximately one-fourth the resiliency of the control specimen compared to approximately three-fourths for the specimen exposed to the liquid phase NTO. In addition, the Viton A specimen exposed to liquid phase INTO-2 was the only specimen which lost weight.

Although the Butyl rubber also exhibited extreme swelling as immediately observed upon removal from the bomb, the swelling was not as great as that of the Viton A and there was no cracking. After exposure to the atmosphere for 30 minutes, the specimens returned to their original size with no apparent color change.

Significantly, there was considerably less reaction between the nonmetals and INTO-2 than between the nonmetals and INTO-1. There was little reaction between INTO-2 and the Butyl rubber but the INTO-1 dissolved nearly all of the Butyl rubber. The amount of remaining inhibitor was also greater during the INTO-2 test. Pretest FNO<sub>2</sub> content was 3.0 mole percent (in the supply bomb) and posttest FNO<sub>2</sub> content was 1.5 mole percent (in the test bomb).

The 321 stainless-steel bomb used during the nonmetals test was prepassivated with an NTO +  $\mathrm{FNO}_2$  solution containing 3.0 mole percent  $\mathrm{FNO}_2$ . The solution remained in the bomb for 24 hours at 23 C. All corrosion test bombs are stored with the valve end down for ease of acquistion of the analysis samples and to eliminate disturbance of the test specimens.

All specimens were weighed after 72 hours and 260 hours exposure to atmosphere. The results, along with appropriate comments, are presented in Tables 5 and 6 and in Fig. 3 through 5. Figure 3 shows the test samples from INTO-1 and NTO, and Fig. 4 shows those from INTO-2 and NTO. Figure 5 shows the residue remaining after the INTO-2 and NTO were allowed to evaporate.

A comparison of the results of the 30-day ambient test with those of the 20-month ambient test should provide the best measure of the effectiveness of  $\mathrm{FNO}_2$  in inhibiting corresivity by NTO. It is to be expected that the 30-day coupons immersed and in contact with the vapor from  $\mathrm{FNO}_2$  + NTO will show as great a change in weight and appearance as those exposed to wet NTO because of the formation of a passivated protective layer. If the layer adequately protects the metals from corrosion, little or no additional change in weight or appearance should occur to the specimens exposed to the  $\mathrm{FNO}_2$  - NTO over the 20-month period whereas the specimens exposed to wet NTO would be expected to be further attacked.

### Galvanic Corrosion

The conductances of wet NTO (0.13 weight percent  $\rm H_20$ ) and INTO-1 (5 weight percent  $\rm F_3N0$ ) prepared from this NTO have been determined as 9 x  $\rm 10^{-12}$  ohm<sup>-1</sup> cm<sup>-1</sup> and 4 x  $\rm 10^{-12}$  ohm<sup>-1</sup> cm<sup>-1</sup> respectively. These two measurements complete the conductance studies on INTO-1 and its components. From the measured values of the conductances, which lie in the region of  $\rm 10^{-11}$  to  $\rm 10^{-12}$  ohm<sup>-1</sup> cm<sup>-1</sup>, it is apparent that galvanic (bimetallic) corrosion will not be a serious problem in the use of INTO-1.

Because INTO-1 is being discarded in favor of INTO-2 it will be necessary to repeat the conductivity tests on INTO-2. These will be carried out during the next quarter.

# SEVEN-DAY COMPATIBILITY TEST OF NONMETALS WITH NTO

Comments		Surface slightly softer in both solutions and both phases; color changed to light amber; no apparent size change	Color changed to light green; no apparent size change	Extreme swelling to 1-3/4 times original size as observed immediately upon removal; surface slightly spongier than control specimen; no apparent color change	No apparent change in size or surface hardness; color changed to buff but returned to original color upon extended exposure to atmosphere (240 hours)	Extreme swelling to approximately twice original size observed upon removal from bomb; reduced to it! original size in 30 minutes upon exposure to stmosphere; greater swelling, cracking, and warping in liquid phase than in vapor phase; slight reduction in elasticity		Color changed to light amber; surface slightly softened; no apparent size change	No apparent size charge; color changed to light green; very slight surface softening	Extreme swelling observed upon removal from bomb; no apparent color change; surface appeared spongler than control specimen	Color changed to buff but returned to normal upon extended exposure to atmosphere (240 hours)	Extreme swelling to about twice original size upon removal from bomb; returned to 1.1 original size after 30-minutes exposure to atmosphere; no apparent color change; slight reduction in elasticity
Percent Weight* Change After 240 Hours		3.5	2.4	5.4	0.2	7.2		3.5	2.5	9*9	0.2	88. 5.
Percent Weight* Change After 72 Hours	- •	7.7	3.7	٠. ٢	0.3	6.7	•	7.7	3.8	9.1	0.3	7.7
Weight Change After 72 Hours, grams	Liquid Phase	0.0233	0.0165	0.0245	0,0018	0.0401	Vapor Phase	0.0236	0.0165	0.0307	0,0016	0.0443
Weight After. grams	Liqu	0.5468	0.4578	0.3428	0.5480	0.6341	Vapo	0.5554	0.4542	0.3690	0.5373	0.6183
Weight Before, grams	_ ,	0.5235	0.4413	0.3183	0.5462	0.5940	•	0.5318	0.4377	0.3383	0.5357	0.5740
Materia.		Kel-F	Kyner	Bubber	Teflon	Viton A		Kel-P	Kynar	Buty! Rubber	Teflon	Viton A

TABLE 6

# SEVEN-DAY COMPATIBILITY TEST OF NONMETALS WITH NTO + FNG $_2$

Corments		Surface slightly softer in both solutions and in both phases after test; color after test, light amber; no apparent size change	Surface slightly softer but not as evident as Kel-F; color changed to light green; no apparent size change	Exhibits a slight sponginess compared to control specimens; characteristics similar in liquid and vapor phase; NTO and INTO-2 extreme swelling observed immediately upon removal (1-3/4 times original size)	No apparent change in surface hardness or size; color changed to buff; afte: extenued exposure to atmosphere, color changed back to original	Extreme swelling to nearly twice original size observed immediately upon removal from test bomb, reduced to 1.1 original size after 30-minutes exposure to amonghere; slightly greater warping, creaking, and reduction in elasticity in DWIO-2 then in NTO		Changed color to light amber; surface slightly softer than control specimen; no apparent change in size	Color changed to light green; no apparent change in size; very slight softening of surface	Extreme swelling observed immediately after removal from bomb; surface apougler than control specimen; no apparent change in color	No apparent change in surface softness or size; color changed to built but returned to original color after extended exposure to atmosphere (240 hours)	Extreme swelling to nearly twice original size observed immediately upon removal from test bomb; reduced to 1.1 original size after 30-minutes exposure to amosphere; reduction in elasticity as in liquid phase; less warping and cracking than in liquid phase; no apparent color 'hange
Percent Weight* Change After 240 Hours		3.6	2.4	4.8	0.07	-2.5		3.6	2.4	0.9	0.1	6.2
Percent Weight* Change After 72 Hours		4.5	3.7	7.1	0.3	4. [-	•	4.5	3.8	æ.	0.3	6.7
Weight Change After 72 Hours, grams	Liquid Phase	0.0240	0.0160	0.0219	9.0014	-0.0079	Vapor Phase	0.0233	0.0164	0.0258	0.0016	0.0407
.t.ght After. grams		0.5562	0.4496	0.3296	0.549н	0.5731	Vel	00.7.00	6.4526	0.3%2H	0.5504	0.6499
Weight Before, grams		0.5322	0.4336	0.3077	0.5484	0.5410		0.736/	0.43%2	0.3170	0.5490	0.1009
Material		Kel-F	Kyner	Butyl Bubber	Tellon	Viton A		Ke1-P	Kyner	Butyl Rubber	Teflon	Viton A

\*Percent weight change based on original weight

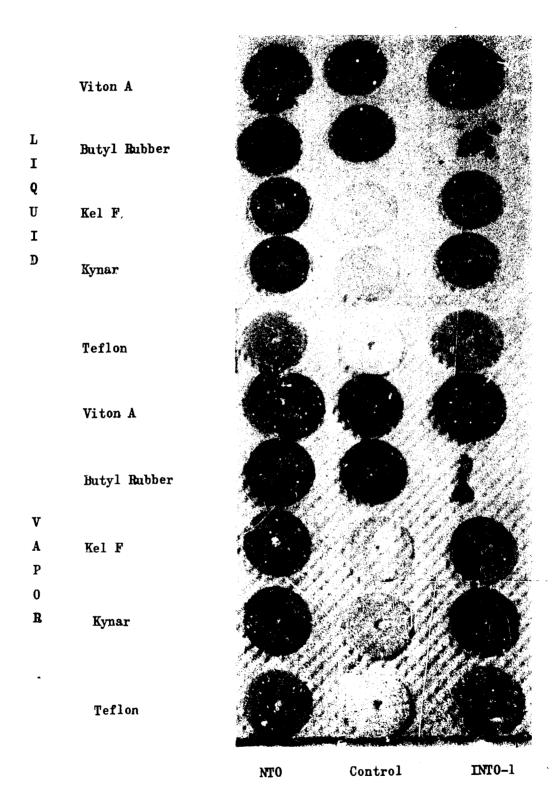


Figure 3. Seven-Day Test Nonmetals

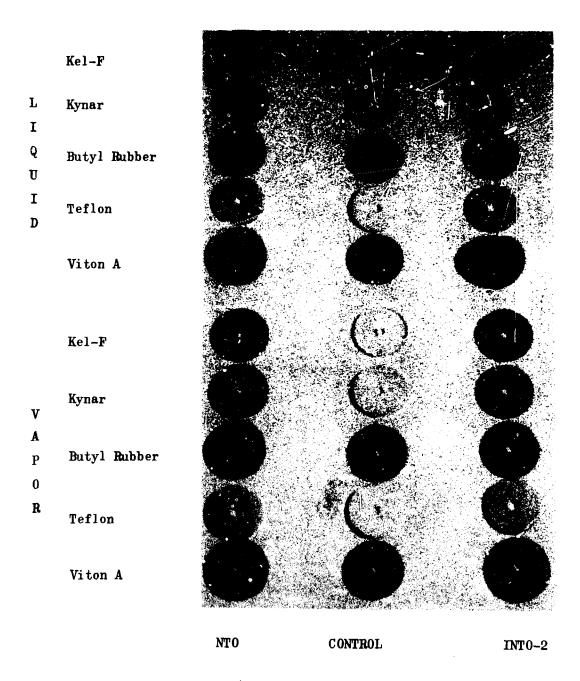


Figure 4. Seven-Day Test Nonmetals



INT0-2

Seven-Day Test Residue

### Storability and Compatibility

INTO-1 was tested for storability and constituents compatibility in 6061 aluminum, 1018 carbon steel, 321 stainless steel, and nickel containers at ambient temperature, 53 C and 70 C. The tests were carried out under two conditions of bomb prepassivation: (1) mild, using  $F_{5}$ NO at 70 C overnight, and (2) vigorous, using  $ClF_{3}$  at 70 C overnight. The bombs had a capacity of 8 to 10 milliliters and each was loaded with 4 to 6 grams of a 4.9-percent  $F_{5}$ NO in NTO mixture. The samples were analyzed by infrared spectrometry and gas chromatography both immediately after loading and after appropriate time periods. The data are summarized in Tables 7 through 9.

An examination of the analytical data immediately reveals that the components of INTO-1 are not compatible at elevated temperatures. As a result, emphasis on the program has been shifted exclusively to INTO-2 (FNO<sub>9</sub> + NTO).

Because INTO-2 was formed by the reaction of F<sub>3</sub>NO with NTO, the compatability tests originally started with INTO-1 were continued using the same solutions in the same bombs. The data from these tests are also contained in Tables 7 through 9. These results indicated that INTO-2 is storable at elevated temperatures, in sufficiently passivated containers, for at least 55 days (45 days at 53 C and 10 days at 70 C). These tests are being continued and the contents of the bombs will be periodically analyzed.

### Vapor Pressure and Solubility

Equally important to determining the corrosivity of INTO on materials of construction is determining the solubility of FNO<sub>2</sub> (boiling point: -72.4 C) in NTO. Preliminary experiments have indicated that the solubility lies within operationally acceptable limits, but exact quantitative data are necessary to design an operational propellant.

TABLE 7

STORABILITY OF INTO-1 AT AMBIENT TEMPERATURE

Material	Mole Percent AMOX at Start	Mole Percent Mole Percent AMOX ALStart After 34 Hours	Mole Percent AMOX	Mole Percent Mole Percent Mole Percent AMOX AMOX AMOX AMOX After 34 Hours After 80 Hours After 160 Hours
Stainless Steel (Working Solenoid Bomb)	4.9			9.4
321 Stainless Steel	6.4	3.4	3.3	2.3
Carbon Steel-1018	3.6	1.6	1.0	1.0
A1-6061	3.9	3.4	3.2	2.8
Nickel	4.2	3.4	3.5	3.2

Passivation: AMOX at 1500 mm absolute pressure, STP, 24 hours at 70 C

Working Solution: 755 grams NTC plus nominal 5 mole percent AMOX in 1-liter stainless-steel high-pressure bomb

STORABILITY AT 53 C

TABLE 8

		Mole Percent F <sub>3</sub> N0	Mole Percent FNO	Mole Percent P <sub>1</sub> NO	Mole Percent FNO <sub>2</sub>	Mole Percent F <sub>1</sub> N0	Mole Percent FNO <sub>2</sub>		
Material	Passivation	at Start	at Start	After 34 Hours After 34 Hours		After 60 Hours	After 80 Hours		
321 Stainless Steel	-	4.0	0	4.0	6.	0	0		
Carbon Steel-1018	1	4.1	0	0.3	2.1	0	0		
A1-6061	1	3.6	•	0.5	•	Trace	•	_	
Nickel	1	3.8	•	0.3	3.3	0	6.4		
				Mole Percent Hole Prrent FyNO PNO After 24 Hours		Mole Percent F <sub>5</sub> NO After 100 Hours	Mole Percent Poly Party FANO FANO After 100 Hours After 100 Hours	Mole Percent Mole Percent FyNO FNO Atter 30 Days After 50 Days	Mole Percent PNO <sub>2</sub> After 30 Days
321 Stainless Steel	81	2.3	0	0.8	0.4	0	4.2	0	4.3
A1-6061	C)	2.7	0	0.7	2.9	0	4.7	•	8°.
Nickel	61	2.5	٥	6.0	3.2	0	4.5	0	4.5
				Mole Percent F <sub>3</sub> NO After 19 Hours	Mole Percent PyNO After 19 Hours Wole Percent After 19 Hours	Mole Percent F <sub>3</sub> NO After 44 Hours	Mole Percent FNO <sub>2</sub> After 44 Hours		
Nickel Body with Monel Valve	r	4·ć	0	1.5	3.8	0.25	4.1		
			_						

Passivation:

2. Three of bombs used with passivation procedure 1 were further passivated with one atmosphere CIFy (STP) at 70 C for 24 hours

3. Single bomb previously used for reaction of 4 grams  $50/50~{
m F}_5{
m NO}$ .NIO mixture at 70 C for 5 days

TABLE 9

STORABILITY AT 70 C

Material	Passivation	Mole Percent $F_{\overline{J}}$ NO at Start	Mole Percent FNO <sub>2</sub> at Start	Mole Percent $\frac{F_{3}N0}{5}$ After 34 Hours	Mole Percent FNO <sub>2</sub> After 34 Hours
321 Stainless Steel	1	0.4	U	0	0
Carbon Steel-1018	7	3.9	0	0	0
A1-6061	7	2.8	0	0	0
Nickel	rd	4.1	0	0	2.5
					After 10 Days
321 Stainless Steel	Q		4.3		4.2
A1-6061	23		8.4		9.4
Nickel	63		4.5		4.5

Passivation: 1 F<sub>3</sub>NO, 1500 mm Hg absolute pressure, STP, 24 hours, 70 C

Three bombs previously passivated as in procedure 1, then further passivated with one atmosphere of CIFz (STP) at 70 C for 24 hours, and stored for 45 days at 53 C; temperature raised to 70 C and storage continued Q

The solubility of FNO<sub>2</sub> in NTO will be determined by the familiar vapor pressure method. For calibration, it will be necessary to check the vapor pressure measurements over the range of 14 to 168 F (critical temperature) on a sample of pure FNO<sub>2</sub>. A precision, temperature-compensated Heise gage will be used for this work.

The vapor pressure and solubility studies proposed for INTO-1 in the previous quarterly report will now be carried out for INTO-2 taking into consideration the lesser volatility and higher critical temperature of FNO<sub>2</sub>.

### Transfer Effects

Because INTO-2 is replacing INTO-1 in this program the investigation of the errosive effects of transfer on materials will be performed with INTO-2 according to the plans delineated in the first quarterly report.

### Venting

The loss of  $FNO_2$  from INTO-2 by venting will be determined as a function of time by the method previously proposed for the loss of  $F_3NO$  from INTO-1.

### RECLAMATION AND PREPARATION STUDIES

Because of the unique method of preparation of INTO-2, the problems of reclamation of NTO and preparation of INTO must be considered together.

INTO-2 was first prepared by the reaction of  $F_3$ NO with NTO. This method of preparation is obviously impractical for large-scale production because of the very high cost of  $F_3$ NO. The method now used to prepare INTO-2 is the reaction of  $F_2$  with NTO. This has been carried out by condensing  $F_2$  into a stainless-steel bomb at -196 C and allowing the mixture to warm to room temperature. This method of preparation has been

reported by Faloon and Kenna Ref. 1. Later in the program attempts will be made to make INTO-2 by adding  $\mathbf{F}_2$  to NTO at ambient temperature, probably by bubbling in the  $\mathbf{F}_2$  If the latter method is successful, it should be possible to design a system for making INTO-2 by adding  $\mathbf{F}_2$  to either in-specification or out-of-specification NTO in field storage tanks.

The problem concerning the presence of out-of-specification NTO in storage tanks has persisted for some time. Sometimes contaminated storage tanks will stand for long period of time before their contents are discarded. The expense of either discarding or reprocessing wet NTO is prohibitive. It might be possible to solve this problem by converting the out-of-specification NTO to INTO by the addition of FNO<sub>2</sub>. Hence, it is very worthwhile to examine the possibility of such a reclamation process.

The feasibility of upgrading NTO with as much as 0.5-percent water equivalent is to be studied during this program.

INTO-2, made both from  $F_3$ NO-NTO and  $F_2$ -NTO, has been shown by Turner bulb military specification analysis performed in a specially constructed aluminum apparatus (Ref. 2) to wrastically reduce the water content originally present in the NTO and also to destroy added water (Table 10). The infrared analysis has shown that a corresponding reduction in  $rNO_2$  content also occurred.

These experiments indicated that reclamation of NTO with 0.3-percent water equivalent is feasible.

### TABLE 10

### EFFECT OF INTO-2 ON WATER EQUIVALENT

<u>0xidizer</u>	water Equivalent, weight percent
NTO	0.15
INT0-2 (from F <sub>2</sub> + NT0)	0.02
INTO-2 (from $F_2$ + NTO after adding 0.15 weight percent $H_0^{0}$ )	0.02

Water Mandanaland

UNUSUAL PHENOMENA OR HAZARDS

Storability-compatibility tests run indicated that INTO-1 was not storable at elevated temperatures. These studies were discussed in the storability section of this report.

These studies confirm the storability problem of  $F_3N0$  in NTO caused by reaction at elevated temperatures. However, the studies also suggest that  $FNO_2$  is a better water scavenger than  $F_3N0$  and is compatible with NTO at elevated temperatures. Because  $FNO_2$  is more readily synthesized (fluorine + NTO), than  $F_3N0$ , its usefulness as a replacement for  $F_3N0$  in inhibited nitrogen tetroxide (INTO) is very promising.

An explosion occurred in a sealed Pyrex u.r.r. sample tube containing neat NTO being prepared for water analysis. The explosion occurred just after the tube had been sealed and the NTO (originally frozen at -196 C) began melting. Some black specks of material were noted in the NTO just prior to the explosion. No injury resulted. It is suspected that some valve packing material may have worked loose and reacted with the NTO.

PHASE II: ANALYTICAL STUDIES

### INFRARED ANALYSIS

On examination of the cylinder of  $F_3N0$  for minor impurities by IR analysis it was found that it contained some  $FN0_2$  which was removed by passing the  $F_3N0$  through a -80 C AlF<sub>3</sub> trap.  $FN0_2$  was also found to be removed by passing the  $F_3N0$  through a 6-foot stainless-steel tube packed with Ascarite.

The thermal stability of INTO-1 was mainly followed by IR analysis and the results are given elsewhere in this report. Becs ise INTO-1 is being replaced by INTO-2 the development of an activate analysis for FNO<sub>2</sub> was necessary. A calibration for FNO<sub>2</sub> has been obtain 1 in the infrared region by using the intensity of the 12.18-micron band of FNO<sub>2</sub> as a quantitative measure of the FNO<sub>2</sub> content. The calibration curve is shown in Fig. 6. This method was used to show that the reaction of F<sub>2</sub> and NTO did produce FNO<sub>2</sub> in sufficient quantity to yield INTO-2. A spectrum of INTO-2 is shown in Fig. 7. Figure 8 shows the calibration curve for the 2.51-micron band of hF which is used to determine the HF which is formed in INTO-2.

### NUCLEAR MAGNETIC RESONANCE 'NALYSIS

The n.m.r. technique for water determination has been used to check the water equivalent of the NTO samples used during the nonmetals and metals corrosion compatability tests and in the galvanic corrosion tests. The NTO used was found to contain less than 0.10 weight percent water equivalent. To prepare wet NTO it will be necessary to add extra water to the available NTO.  $F^{19}$  n.m.r. confirmed the formation of FNO<sub>2</sub> by the reaction between  $F_2$  and NTO.

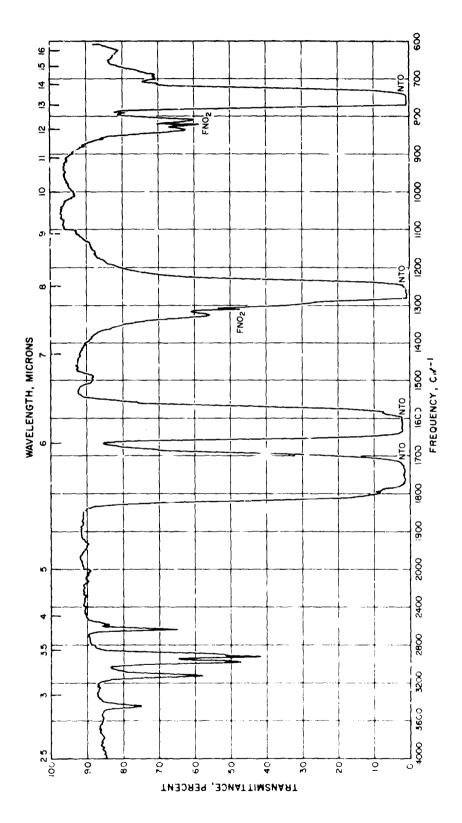


Figure 6. Infrared Calibration Curve

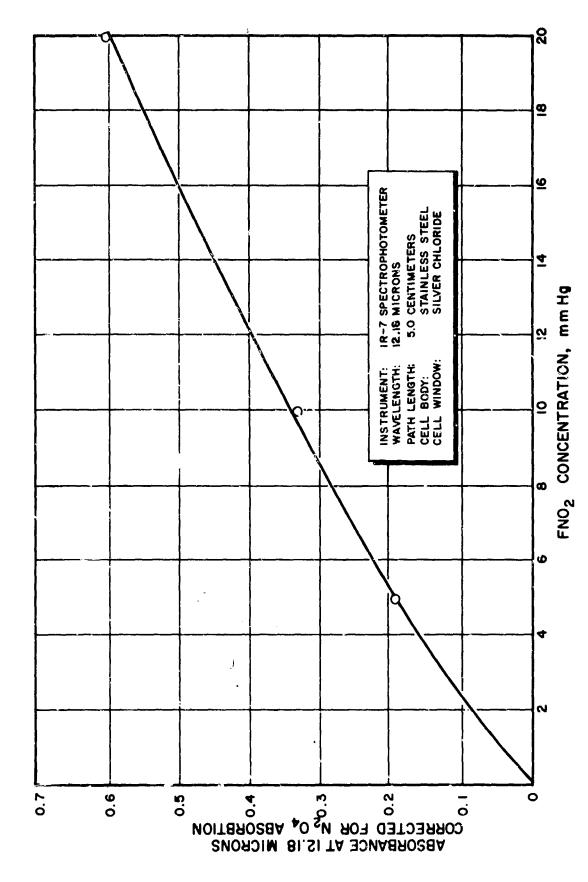
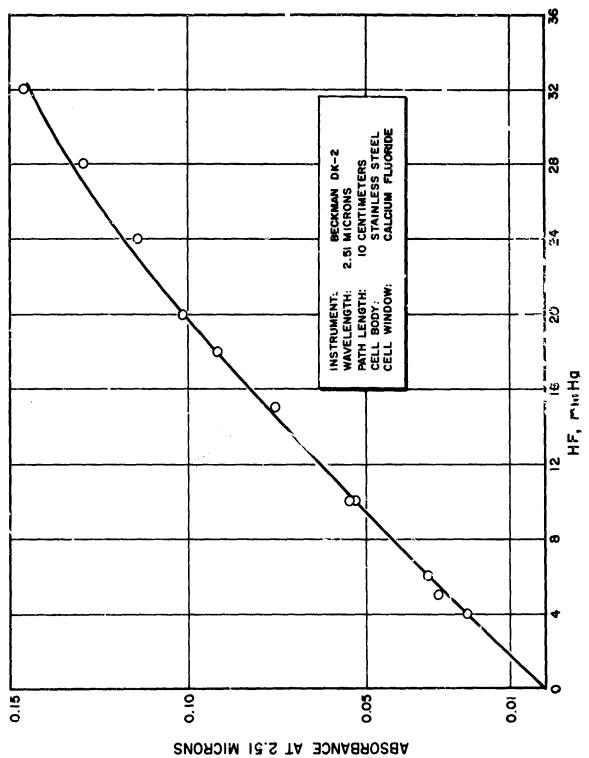


Figure 7. Spectrophotometer Calibration Curve for  ${
m FNO}_2$  in  ${
m N}_2{
m O}_4$ 





Spectrophote meter Calibration Curve for HF Figure 8.

### WATER EQUIVALENT TESTING

The all-aluminum model of the military specification water equivalency apparatus was used on a sample of NTO tested by the n.m.r. method. Using the aluminum apparatus, the water equivalent was found to be 0.21 weight percent; using n.m.r.,.13 weight percent; and using the glass military specification apparatus, 0.15 weight percent.

After using the aluminum apparatus for this determination, a considerable amount of sludge was found in the bulb. A sample of INTO-1 made from the above NTO had a water equivalent of 0.04 and 0.05 weight percent as determined by the glass military specification apparatus. These latter results may be slightly high because of reaction of HF with the glass.

The aluminum apparatus was then passivated with 48-percent HF in  $\rm H_2^{}0$ . The NTO analyzed using this apparatus was found to contain 0.15 weight percent water equivalent but this was reduced to 0.03 weight percent when the NTO was converted to INTO-2 by the addition of  $\rm F_2$ . To this sample of INTO-2 water was added equivalent to 0.2 weight percent; the water equivalent was measured and found to still be 0.03 weight percent. No evidence was found of the sludge which had been produced by reaction of NTO with the same apparatus prior to passivation with HF.

### FUTURE EFFORT

The galvanic corrosion studies will be repeated for INTO-2. The loading of samplers for short- and long-term corrosivity tests will be completed and the tests will be continued. Work will proceed on the vapor pressure, solubility, and freezing point studies. Reclamation studies will be carried out in conjunction with improved practical methods of preparing INTO-2 from fluorine and NTO. Analytical methods will be improved or modified as required to support the program.

### REFERENCES

- 1. Falcon, A. V. and W. B. Kenna, J. Am. Chem. Soc., 73, 2937 (1951).
- 2. R-6354-1, Inhibited N<sub>2</sub>O<sub>4</sub>, First Quarterly Progress Report, Rocketdyne, a Division of North American Aviation, Inc., Canoga Park, California, October 1965. CONFIDENTIAL

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